



Sharif University of Technology  
**Scientia Iranica**  
*Transactions B: Mechanical Engineering*  
www.scientiairanica.com



# Thermally induced vibration of an electro-statically deflected functionally graded micro-beam considering thermo-elastic coupling effect

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Received 19 December 2012; received in revised form 26 September 2013; accepted 16 December 2013

## KEYWORDS

Thermally induced vibration;  
FGM;  
Thermo-elastic coupling;  
Electrostatic load.

**Abstract.** This study investigates the dynamic response of an electrostatically deflected capacitive cantilever Functionally Graded (FG) micro-beam subjected to a harmonically varying thermal load, which incites vibration due to the different material properties of the beam constituents and the thermo-elastic coupling effect. The FG beam is made of a mixture of metal and ceramic, where the material properties vary continuously through the thickness according to an exponential distribution law (E-FGM). Assuming the Euler-Bernoulli beam theory and the infinite speed of heat transportation, the equation of motion and the conventional coupled energy equation are derived. Applying Galerkin formulation and then using the Rung-kutta method as an efficient numerical tool, these equations are simultaneously solved. Changing the ceramic constituent percentage of the bottom surface, five different types of FGM micro-beams are investigated and the results are presented for all types. Numerical results show the response of a cantilever FG micro-beam subjected to a harmonically varying temperature input. Moreover, the influences of the beam ceramic constituent percentage on stability, vibrational behavior and natural frequency are presented.

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## 1. Introduction

A new generation of advanced composites includes Functionally Graded Materials (FGMs). These materials have a smooth variation of material properties along one or more directions. Better stress relaxation and the potential to resist large temperature gradients are the main advantages of this class of materials. Typically, FGMs consisting of thermal-resisting ceramic and fracture-resisting metal are composites whose volume

fractions of constituents vary gradually, resulting in a continuous change in physical properties [1-3].

The study of the dynamic behavior of FG structures under mechanical or thermal loads certainly has been an active subject of research, as evidenced by the many analytical and numerical analyses published in literature. Numerous researchers have studied the vibration and heat transfer process of these kinds of beam. Sankar and his co-workers [4,5] carried out the elasticity solutions of thermally loaded FG beams. By considering shear deformation, Chakraborty et al. [6] presented a new beam element to study the thermo-elastic behavior of FG beam structures. Using the Meshless Local Petrov-Galerkin method (MLPG), Qian and Ching [7] obtained the numerical solutions for static deformation, and free and force vibrations of a FG cantilever beam, while Ching and Yen [8]

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analyzed the 2D FG solids, which are subjected to either mechanical or thermal loads. By means of different higher order shear deformation theories and the Classical Beam Theory (CBT), the free vibration of a FG beam is presented by Aydogdu and Taskin [9]. Their results showed that CBT gives higher results and, also, they claimed that vibration frequencies, mode shapes and dynamic response are considerably changed by thickness variation, temperature change, slenderness ratio, volume fraction index, FG layer thickness, and end support conditions. Based on the Timoshenko beam theory, Xiang and Yang [10] explored both free and forced vibration of an FG beam with changeable thickness under thermally induced initial stresses. Kapuria et al. [11] proposed a finite element model for static and free vibration responses of layered FG beams using a third-order theory and its experimental validation. Li [12] presented a new unified approach to investigate the static and free vibration behavior of Euler-Bernoulli and Timoshenko beams. Using the modified differential quadrature method, Pradhan and Murmu [13] studied the thermo-mechanical vibration analysis of FGM beams and sandwich beams resting on a variable Winkler foundation. Abbasi et al. [14] used a Galerkin based finite element method to analyze the vibration behavior of a FG Timoshenko beam under lateral thermal shock with coupled thermo-elastic assumption, and, similarly, Babaei et al. [15] studied the thermo-elastic vibration of FG beams under lateral thermal shock with the Euler-Bernoulli beam assumptions. Sina et al. [16] developed a new beam theory, different from the traditional first-order shear deformation beam theory, with the purpose of analyzing the free vibration of FG beams considering different boundary conditions and making comparisons between various beam theories. Mahi et al. [17] presented an exact solution to study the free vibration and frequency response of a beam made of symmetric FG materials. In their work, the beam is assumed to be initially stressed by a temperature rise through the thickness. Free and force vibration analysis of FG beams considering the temperature dependency of material properties is discussed by Azadi [18]. In his work, the material properties were assumed to be temperature-dependent. The two-dimensional solution for a simply supported beam made of FG material subjected to arbitrary time-dependent lateral thermal shock loads, using the semi-analytical finite element method, was undertaken by Afshar et al. [19]. The equation of motion and the conventional coupled energy equation are simultaneously solved to obtain the displacement components and temperature distribution in the beam. Malekzadeh and Shojaee [20] carried out the dynamic response of FG beams under a moving heat source considering the material properties to be temperature-dependent. A two-dimensional Finite

Element Method (FEM) is employed to obtain the temperature distribution throughout the beam. Then, the effect of a two-dimensional variation of temperature on the dynamic response of an FG beam with arbitrary boundary conditions is formulated, based on the first-order shear deformation beam theory (FSDBT). The dynamic stiffness method is developed to investigate the free vibration behavior of FG beams by Su et al. [21]. Hemmatnezhad et al. [22] analyzed the large-amplitude free vibration of FG beams through a finite element formulation. They studied the influences of the power-law exponent, vibration amplitude, beam geometrical parameters and end supports on the free vibration frequencies.

Recently, the emerging technology of micro- and nano-electromechanical systems (MEMS and NEMS) has seen dramatic progress in fabricating and testing new devices, creating innovative applications, and proposing novel technologies. Their light weight, small size, low-energy consumption and durability make them even more attractive. Accurate analysis of various effects on the characteristics of resonators, such as resonant frequencies and quality factors, is crucial for designing high-performance components [23]. Many authors have studied the vibration and heat transfer process of micro- and nano-beams. Laser-induced vibration of micro-beams is investigated by Sun et al. [24,25] considering a two dimensional hyperbolic heat conduction model with one relaxation time. Applying different boundary conditions, they developed the coupled thermo-elastic vibration of a micro scale beam resonator induced by pulsed laser heating. Mohammadi-Alasti et al. [26] studied the static behavior of FG cantilever micro-beams and their static instability, when the beam is subjected to a nonlinear electrostatic pressure and temperature changes. They derived nonlinear integro-differential thermo-electromechanical equations based on the Euler-Bernoulli beam theory. Most recently, the static and dynamic behavior of the FG micro-beam, based on Modified Couple Stress Theory (MCST), subjected to nonlinear electrostatic pressure and thermal changes, regarding convection and radiation, is investigated by Zamanzadeh et al. [27]. Youssef and Elsibai [28] discussed the vibration of a gold nano-beam induced by different types of thermal loading. Also, in another work, Youssef [29] carried out the solution of vibration of a gold nano-beam resonator induced by thermal shock in the context of generalized thermoelasticity with variable thermal conductivity.

Many attempts have been made to study the vibration and heat transfer process of beams. The above-mentioned articles reveal that there has been widespread study of the vibrational behaviour of both homogenous and FG beams under either mechanical or thermal loads. More recently, there has been

investigation into thermally induced vibration of (only) pure metal micro- and nano-beams. To the best of the authors' knowledge, no previous work has been done concerning the coupled thermoelastic vibrational behavior of a micro beam made of FG materials, which is subjected to not only electrostatic pressure, but also harmonically varying thermal load. It must be pointed out that the problem of sinusoidal heat source can arise when a dynamic remote capacitive temperature sensor is used to measure rotating hot objects such as turbine blades. In this study, using the Galerkin based discretizing method, the coupled thermo-elastic vibration of a capacitive FG Euler-Bernoulli micro-beam induced by harmonic thermal loads is investigated. Moreover, the effect of ceramic constituent percentage on vibration and natural frequency is determined and results are considered to highlight the behavioral difference of FG material micro beams with pure metal ones.

## 2. Model description and mathematical modeling

### 2.1. Material properties

The schematic diagram of the FG micro-beam is presented in Figure 1. The neutral axis and coordinates of the composite beam are shown in this figure, too. The studied model is a cantilever capacitive FG micro-beam with a rectangular cross section, with dimensions of length  $L(0 \leq x \leq L)$ , width  $b(-b/2 \leq y \leq b/2)$  and thickness  $h(-h/2 \leq z \leq h/2)$ . The  $h/L$  ratio is assumed to be small enough to eliminate the shear deformation effects. We define the  $x$  axis along the axis of the beam, and the  $y$  and  $z$  axes correspond to the width and thickness, respectively. In equilibrium, the beam is unstrained and unstressed.

In this study, the material properties of the FG beam are assumed to vary through the thickness according to an exponential law function. Assuming  $\bar{z} = z + h/2$ , the exponential law is given by [27]:

$$P(z) = P_t e^{\delta_0 \bar{z}} = P_t e^{\delta_0 (z+h/2)}, \quad \delta_0 = \frac{1}{h} \ln \left( \frac{P_b}{P_t} \right). \quad (1)$$

$P(z)$  denotes a typical material property.  $P_t$  and  $P_b$  denote the values of the properties at the top

and bottom of the FG beam. It is assumed that the top surface is made of pure metal ( $P_t =$  metal properties), but the bottom surface from a mixture of metal and ceramic. Also, it is assumed that the ceramic constituent percentage of the bottom surface varies from 0% to 100%. In order to determine the material properties of the bottom surface ( $P_b$ ), a volume fraction of material is used [27]:

$$P_b = V_c P_c + V_m P_m, \quad (2)$$

where  $V$  is the volume fraction, and subscripts, “ $m$ ” and “ $c$ ”, stand for metal and ceramic, respectively. So,  $V_m$  and  $V_c$  are the volume-fractions of the metal and ceramic, respectively, and are related by:

$$V_m + V_c = 1. \quad (3)$$

Changing the ceramic constituent percentage of the bottom surface, five different types of FG micro-beam are investigated. Using the above mentioned equations, parameter  $\delta_0$  is specified for some material properties, as follows:

$$\begin{aligned} \gamma &= \frac{1}{h} \ln \left( \frac{E_b}{E_m} \right), & \beta &= \frac{1}{h} \ln \left( \frac{\alpha_b}{\alpha_m} \right), \\ \lambda &= \frac{1}{h} \ln \left( \frac{K_b}{K_m} \right), & \mu &= \frac{1}{h} \ln \left( \frac{\rho_b}{\rho_m} \right), \\ \eta &= \frac{1}{h} \ln \left( \frac{C_{vb}}{C_{vm}} \right), & \mathcal{S} &= \frac{1}{h} \ln \left( \frac{v_b}{v_m} \right), \end{aligned} \quad (4)$$

in which  $E$ ,  $\alpha$ ,  $K$ ,  $\rho$ ,  $C_v$  and  $v$  denote the Young's modulus of elasticity, thermal expansion coefficient, thermal conductivity, density, specific heat and Poisson's ratio, respectively.

### 2.2. Dynamic equation of motion

According to the basic hypotheses of Euler-Bernoulli beams and the one-dimensional beam theory, the displacement field for a FG beam can be written as [30]:

$$u = u_0 - z \frac{\partial W}{\partial x}, \quad \nu = 0, \quad W = W(x, t), \quad (5)$$

where  $u_0$  is the axial displacement of the beam in the reference plane and  $u$ ,  $\nu$  and  $W$  are the  $x$ ,  $y$  and  $z$

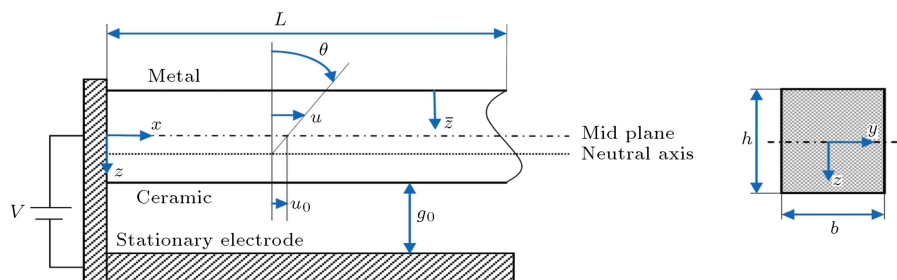


Figure 1. Geometry and coordinates of symmetric capacitive FG micro-beam (side and section view).

components of the displacement vector, respectively. The axial strain of the beam, based on the Euler-Bernoulli beam theory, is expressed as [30]:

$$\varepsilon_x = \frac{du}{dx} = \frac{\partial u_0}{\partial x} - z \frac{\partial^2 W}{\partial x^2}. \tag{6}$$

The resultant stress, considering temperature changes for the Euler-Bernoulli beam, in a cross section area of the micro-beam, using Hooke's law, can be presented as [30]:

$$\begin{aligned} \sigma_x &= \bar{E}(z) (\varepsilon_x - \bar{\alpha}(z)\theta(x, z, t)) \\ &= \bar{E}(z) \left( \frac{\partial u_0(x, t)}{\partial x} - z \frac{\partial^2 W(x, t)}{\partial x^2} \right) \\ &\quad - \bar{E}(z)\bar{\alpha}(z)\theta(x, z, t), \end{aligned} \tag{7}$$

where  $\theta = T(x, z, t) - T_\infty$  is the temperature increment of the beam, measured with respect to a reference (environment) temperature ( $T_\infty$ ). It is worth pointing out that for plain stress conditions,  $\bar{E}$  and  $\bar{\alpha}$  are equal to  $E$  and  $\alpha$ , respectively, and for plain strain condition, are  $E/(1 - \nu^2)$  and  $1 + \nu$ , respectively [31]. Under no axial force, the equilibrium equations of the cantilever FG micro-beam are considered as:

$$\int_A \sigma_x dA = \int_A E(z) \left( \frac{\partial u_0(x, t)}{\partial x} - z \frac{\partial^2 W(x, t)}{\partial x^2} - \alpha(z)\theta(x, z, t) \right) dA = 0, \tag{8}$$

$$\int_A \sigma_x z dA = \int_A z E(z) \left( \frac{\partial u_0(x, t)}{\partial x} - z \frac{\partial^2 W(x, t)}{\partial x^2} - \alpha(z)\theta(x, z, t) \right) dA = M. \tag{9}$$

As a result, from Eq. (8),  $\frac{\partial u_0(x, t)}{\partial x}$  can be determined as:

$$\frac{\partial u_0(x, t)}{\partial x} = \frac{1}{\bar{A}} \left( \bar{B} \frac{\partial^2 W}{\partial x^2} + F_T \right). \tag{10}$$

The parameters appeared in Eq. (10) are:

$$\bar{A} = \int_{-h/2}^{h/2} e^{\gamma(z+h/2)} dz, \tag{11}$$

$$\bar{B} = \int_{-h/2}^{h/2} z e^{\gamma(z+h/2)} dz,$$

$$F_T = \int_{-h/2}^{h/2} \alpha_m e^{(z+h/2)(\gamma+\beta)} \theta(x, z, t) dz.$$

Substituting  $\frac{\partial u_0(x, t)}{\partial x}$  into the moment equilibrium equation (Eq. (9)), the motion equation of a FG beam subjected to an external heat source may be obtained as:

$$-(EI)_{eq} \frac{\partial^2 W}{\partial x^2} = M - M_T, \tag{12}$$

in which  $M$  is the external bending moment in a given section produced from the electrostatic force acting along the micro-beam and  $W$  is the transverse deflection.  $(EI)_{eq}$  and  $M_T$  are the equivalent bending stiffness and thermal moment, respectively:

$$(EI)_{eq} = \int_A \left( z E(z) \left( \frac{\int_A z E(z) dA}{\int_A E(z) dA} \right) - z^2 E(z) \right) dA, \tag{13}$$

$$\begin{aligned} M_T(x, t) &= \int_A \left( \frac{z}{\int_A E(z) dA} E(z) F_T \right. \\ &\quad \left. - z E(z) \alpha(z) \theta(x, z, t) \right) dA. \end{aligned} \tag{14}$$

The governing motion equation of the transversal displacement of a FG micro-beam,  $W(x, t)$ , actuated by an electrical load and thermal moment, is written as [31-25]:

$$\begin{aligned} (EI)_{eq} \frac{\partial^4 W(x, t)}{\partial x^4} + \frac{\partial^2 M_T(x, z, t)}{\partial x^2} \\ + (\rho A)_{eq} \frac{\partial^2 W(x, t)}{\partial t^2} + c \frac{\partial W(x, t)}{\partial t} = F(W, V). \end{aligned} \tag{15}$$

The micro-beam may be assumed subjected to a viscous damping, which can be due to squeeze-film damping. This effect is approximated by an equivalent external damping coefficient per unit length ( $c$ ) [23]. Since the micro beam is assumed to be subjected to a bias DC voltage, an electrostatic distributed force is introduced as [31]:

$$F(W, V) = \frac{\varepsilon_0 b V^2}{2(g_0 - W)^2}, \tag{16}$$

where  $\varepsilon_0$  is the dielectric coefficient (permittivity) of the air,  $b$  is the width of the beam, and  $g_0$  is the initial gap between the micro-beam and the ground electrode. Parameter  $(\rho A)_{eq}$ , appeared in Eq. (15), equals:

$$(\rho A)_{eq} = \int_A \rho(z) dA = b \int_{-h/2}^{h/2} \rho_t e^{\mu(z+h/2)} dz. \tag{17}$$

The small vibration of an electrostatically deflected micro-beam can be studied by introducing the dynamic

deflection of the micro-beam about the static equilibrium position,  $W_s(x)$  as  $W_d(x, t)$ , therefore, the total deflection can be expressed as [31]:

$$W(x, t) = W_s(x) + W_d(x, t), \tag{18}$$

$$W_d(x, t) \ll W_s(x).$$

Using the Calculus of Variation Theory and Taylor series expansion about  $W_d(x, t)$ , and neglecting higher order terms, the linearized dynamic motion equation is obtained as:

$$(EI)_{eq} \frac{\partial^4 W_d}{\partial x^4} + \frac{\partial^2 M_T}{\partial x^2} + (\rho A)_{eq} \frac{\partial^2 W_d}{\partial t^2} + c \frac{\partial W_d}{\partial t} - \frac{\varepsilon_0 b V^2}{(g_0 - W_s)^3} W_d = 0. \tag{19}$$

**2.3. Equation of thermal distribution**

According to literature, until now, the classical Fourier heat conduction theory has been widely used for the study of micro-beams [25]. To determine temperature distribution into the beam, the first law of thermodynamics for heat conduction in the coupled form, assuming the infinite speed of heat transportation, is used [15]:

$$(k\theta_{,i})_{,i} - \rho(z)C_\nu(z)\theta_{,t} - \alpha(z) (3\bar{\lambda} + 2\bar{\mu}) T_\infty \dot{\varepsilon}_{ii} = 0, \tag{20}$$

where  $\varepsilon_{ii}$  is the trace of the strain tensor, and  $\bar{\lambda}$  and  $\bar{\mu}$  are Lamé constants. Temperature variation may assume to occur in the thickness and  $x$  directions, and the temperature field is considered constant in the  $y$  direction. Therefore, the energy equation for the beam under consideration is reduced to:

$$k(z) \frac{\partial^2 \theta}{\partial z^2} + \frac{\partial k(z)}{\partial z} \frac{\partial \theta}{\partial z} + k(z) \frac{\partial^2 \theta}{\partial x^2} - \rho(z)C_\nu(z) \frac{\partial \theta}{\partial t} - \alpha(z) (3\bar{\lambda} + 2\bar{\mu}) T_\infty \dot{\varepsilon}_{ii} = 0. \tag{21}$$

Based on the Euler-Bernoulli beam theory, the trace of the strain tensor of a FG beam for plane stress conditions is [30]:

$$\varepsilon_{ii} = (1 - 2\nu(z)) \left( \frac{\partial u_0}{\partial x} - z \frac{\partial^2 W}{\partial x^2} \right) + 2(1 + \nu(z)) \alpha(z) \theta(z). \tag{22}$$

Substituting  $\frac{\partial u_0(x,t)}{\partial x}$  from Eq. (10) into Eq. (22), it follows:

$$\dot{\varepsilon}_{ii} = (1 - 2\nu(z)) \left( \left( \frac{\tilde{B}}{\tilde{A}} - z \right) \frac{\partial^3 W_d}{\partial t \partial x^2} + \frac{1}{\tilde{A}} \frac{\partial F_T}{\partial t} \right) + 2(1 + \nu(z)) \alpha(z) \frac{\partial \theta(x, z, t)}{\partial t}. \tag{23}$$

Finally, using Eq. (21) and substituting Eq. (23) and Lamé constants into Eq. (21), the energy equation takes the following form:

$$k(z) \frac{\partial^2 \theta}{\partial x^2} + \frac{\partial k(z)}{\partial z} \frac{\partial \theta}{\partial z} + k(z) \frac{\partial^2 \theta}{\partial z^2} = \left( \rho(z)C_\nu(z) + \frac{2(1 + \nu(z))}{(1 - 2\nu(z))} E(z)\alpha(z)^2 T_\infty \right) \frac{\partial \theta}{\partial t} + E(z)\alpha(z) T_\infty \left\{ \left( \frac{\tilde{B}}{\tilde{A}} - z \right) \left[ \frac{\partial^3 W_d}{\partial t \partial x^2} \right] + \frac{1}{\tilde{A}} \frac{\partial F_T}{\partial t} \right\}. \tag{24}$$

For convenience in analysis, the following dimensionless parameters are presented to transform Eqs. (19) and (24) into non-dimensional forms:

$$\widehat{W}_d = \frac{W_d}{g_0}, \quad \widehat{W}_s = \frac{W_s}{g_0}, \quad \hat{x} = \frac{x}{L}, \tag{25}$$

$$\hat{z} = \frac{z}{h}, \quad \hat{t} = \frac{t}{t^*}, \quad \hat{c} = \frac{cL^4}{(EI)_{eq} t^*},$$

$$t^* = L^2 \sqrt{\frac{(\rho A)_{eq}}{(EI)_{eq}}}, \quad \widehat{M}_T = \frac{L^2 M_T}{(EI)_{eq} g_0}.$$

Finally, by applying these non-dimensional parameters, the thermo-elastic coupled equations are rearranged into the following equations:

$$\frac{\partial^4 \widehat{W}_d}{\partial \hat{x}^4} + \frac{\partial^2 \widehat{M}_T}{\partial \hat{x}^2} + \frac{\partial^2 \widehat{W}_d}{\partial \hat{t}^2} + \hat{c} \frac{\partial \widehat{W}_d}{\partial \hat{t}} - \mathcal{A}_1 \frac{V^2}{(1 - \widehat{W}_s)^3} \widehat{W}_d = 0, \tag{26}$$

$$k(\hat{z}) \frac{\partial^2 \theta}{\partial \hat{z}^2} + \frac{\partial k(\hat{z})}{\partial \hat{z}} \frac{\partial \theta}{\partial \hat{z}} + \mathcal{A}_2 k(\hat{z}) \frac{\partial^2 \theta}{\partial \hat{x}^2} - \mathcal{A}_3 \left( \rho(\hat{z})C_\nu(\hat{z}) + \frac{2(v(\hat{z}) + 1)}{1 - 2v(\hat{z})} E(\hat{z})\alpha(\hat{z})^2 T_\infty \right) \frac{\partial \theta}{\partial \hat{t}} - [-\mathcal{A}_4 \hat{z} + \mathcal{A}_5] E(\hat{z})\alpha(\hat{z}) \frac{\partial^3 \widehat{W}_d}{\partial \hat{t} \partial \hat{x}^2} - \mathcal{A}_6 E(\hat{z})\alpha(\hat{z}) \frac{\partial \widehat{F}_T}{\partial \hat{t}} = 0, \tag{27}$$

where:

$$\mathcal{A}_1 = \frac{\varepsilon b l^4}{(EI)_{eq} g_0^3}, \quad \mathcal{A}_2 = \frac{h^2}{L^2},$$

$$\mathcal{A}_3 = \frac{h^2}{t^*}, \quad \mathcal{A}_4 = T_\infty \frac{h^3 g_0}{t^* L^2},$$

$$\mathcal{A}_5 = T_\infty \frac{\tilde{B} h^2 g_0}{\tilde{A} t^* L^2}, \quad \mathcal{A}_6 = \frac{h^2 T_\infty}{\tilde{A} t^*}. \tag{28}$$

The corresponding boundary conditions are:

$$\begin{aligned} \widehat{W}_d(0, \hat{t}) = 0, \quad \frac{\partial \widehat{W}_d}{\partial \hat{x}}(0, \hat{t}) = 0, \\ -(EI)_{eq} \frac{\partial^2 W_d}{\partial x^2}(L, t) = -M_T(L, t) \\ \rightarrow \frac{\partial^2 \widehat{W}_d}{\partial \hat{x}^2}(1, \hat{t}) = \frac{L^2 M_T}{(EI)_{eq} g_0}, \\ -(EI)_{eq} \frac{\partial^3 W_d}{\partial x^3}(L, t) = -\frac{\partial M_T(L, t)}{\partial x} \\ \rightarrow \frac{\partial^3 \widehat{W}_d}{\partial \hat{x}^3}(1, \hat{t}) = \frac{L^2}{(EI)_{eq} g_0} \frac{\partial M_T(1, \hat{t})}{\partial \hat{x}}. \end{aligned} \quad (29)$$

The thermal boundary conditions can be assumed time varying in the upper and lower surfaces but isolated at its free end. The biot number ( $hL/K$ ) is so small (about  $10^{-8}$ ) at micro scale that the temperature of the body is approximately uniform when temperature distribution has an external source [32]. But, in this case, the temperature distribution has external and internal sources due to thermo-elastic coupling effects and, therefore, the temperature changes ( $\theta(x, z, t)$ ) include two terms. The term due to the external heat source is considered uniform, because of the small biot number, but the term due to the internal heat source has a distribution over the beam domain.

$$\theta(x, z, t) = \mathcal{T}(t) + \Psi(x, z, t), \quad \mathcal{T}(t) = \sin(\omega t), \quad (30)$$

in which  $\omega$  is the frequency of the external heat source. The beam is initially assumed to be at ambient temperature ( $\theta(x, z, t) = 0$ ).

### 3. Numerical solutions

#### 3.1. Static deflection equation

Static deflection of the electrostatically deflected micro-beam can be obtained via solving Eq. (15) by neglecting inertial terms. The considered method to solve the equation of the static deflection consists of two steps. In the first stage, a step by step linearization method (SSLM) is used and, in the second, a Galerkin method is used, for solving the linear obtained equation. The SSLM is a method for resolving the problems associated with the non-linearity, changing the governing equation into a linear one. Assume  $V^i$  is the applied voltage, which causes a deflection in the micro-beam ( $\widehat{W}_s^i$ ) in the  $i$ th step. An increment in the applied voltage (denoted by  $\delta V$ ) in the  $(i+1)$ th step results in an increase in the beam deflection represented by [26]:

$$\begin{aligned} V^{i+1} = V^i + \delta V, \quad \widehat{W}_s^{i+1} = \widehat{W}_s^i + \delta W, \\ \delta W = \Gamma(\hat{x}), \quad i = 1 : N. \end{aligned} \quad (31)$$

Considering a small value of  $\delta V$ , the value of  $\Gamma(\hat{x})$  will be expected to be small enough to obtain a desired accuracy. The equation of the FG micro-beam at the  $(i+1)$ th step can be expressed as:

$$\frac{\partial^4 \widehat{W}_s^{i+1}(\hat{x})}{\partial \hat{x}^4} = \mathcal{J} \frac{(V^{i+1})^2}{(1 - \widehat{W}_s^{i+1}(\hat{x}))^2}. \quad (32)$$

Parameter  $\mathcal{J}$ , appeared in Eq. (32), equals  $\frac{\varepsilon b l^4}{2(EI)_{eq} g_0^3}$ . Using the Calculus of Variation Theory and Taylor series expansion about  $\widehat{W}_s^i$  and  $V^i$ , and also neglecting its higher orders, Eq. (32) is rearranged to:

$$\frac{\partial^4 \Gamma(\hat{x})}{\partial \hat{x}^4} - 2\mathcal{J} \frac{(V^i)^2}{(1 - \widehat{W}_s^i)^3} \Gamma(\hat{x}) - 2\mathcal{J} \frac{V^i}{(1 - \widehat{W}_s^i)^2} \delta V = 0. \quad (33)$$

The obtained linear equation can be solved using a Galerkin based weighted residual method. By choosing suitable shape functions,  $\zeta_i(\hat{x})$ , and satisfying the geometrical boundary conditions of the micro-beam,  $\Gamma(\hat{x})$  can be approximated by the following series:

$$\Gamma(\hat{x}) = \sum_{i=1}^N a_i \zeta_i(\hat{x}), \quad (34)$$

where  $a_i$  are constant coefficients, which are calculated at each step.

#### 3.2. Coupled thermo-elastic equation

The obtained generalized coupled governing differential equations with time varying boundary conditions can be transformed into an enhanced form with homogeneous boundary conditions. Due to the time varying nature of boundary conditions (Eqs. (29) and (30)),  $\widehat{W}_d(\hat{x}, \hat{t})$  and  $\theta(\hat{x}, \hat{z}, \hat{t})$  are introduced as:

$$\widehat{W}_d(\hat{x}, \hat{t}) = \widehat{W}_{0T}(\hat{x}, \hat{t}) + \varpi(\hat{x}, \hat{t}),$$

$$\begin{aligned} \widehat{W}_{0T}(\hat{x}, \hat{t}) = \frac{L^2}{(EI)_{eq} g_0} \left[ \frac{\partial M_T}{\partial \hat{x}} \Big|_{(1, \hat{t})} \frac{\hat{x}^3}{6} \right. \\ \left. + \left( M_T(1, \hat{t}) - \frac{\partial M_T}{\partial \hat{x}} \Big|_{(1, \hat{t})} \right) \frac{\hat{x}^2}{2} \right], \end{aligned} \quad (35)$$

$$\theta(\hat{x}, \hat{z}, \hat{t}) = \mathcal{T}(\hat{t}) + \Phi(\hat{x}, \hat{z}, \hat{t}). \quad (36)$$

Parameters  $\varpi$ ,  $\Phi$  and  $M_T(1, \hat{t})$  are introduced later. The accompanying dimensionless homogenous boundary conditions are:

$$\begin{aligned} \varpi(0, \hat{t}) = 0, \quad \frac{\partial \varpi}{\partial \hat{x}}(0, \hat{t}) = 0, \\ \frac{\partial^3 \varpi}{\partial \hat{x}^3}(1, \hat{t}) = 0, \quad \frac{\partial^2 \varpi}{\partial \hat{x}^2}(1, \hat{t}) = 0, \end{aligned} \quad (37)$$

$$\begin{aligned} \Phi(\hat{x}, -\frac{1}{2}, \hat{t}) &= 0, \quad \Phi(\hat{x}, \frac{1}{2}, \hat{t}) = 0, \\ \Phi(0, \hat{z}, \hat{t}) &= 0, \quad \frac{\partial \Phi}{\partial x}(1, \hat{z}, \hat{t}) = 0. \end{aligned} \tag{38}$$

Based on the Galerkin based reduced order model, the deflection and temperature changes of the beam can be approximated in terms of linear combinations of a finite number of suitable shape functions with time dependent coefficients:

$$\begin{aligned} \varpi(\hat{x}, \hat{t}) &= \sum_{k=1}^p U_k(\hat{t})\psi_k(\hat{x}) \rightarrow \\ \hat{W}_d(\hat{x}, \hat{t}) &= \hat{W}_{0T} + \sum_{k=1}^p U_k(\hat{t})\psi_k(\hat{x}), \tag{39} \\ \Phi(\hat{x}, \hat{z}, \hat{t}) &= \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t})\varphi_i(\hat{x})\phi_j(\hat{z}) \rightarrow \\ \theta(\hat{x}, \hat{z}, \hat{t}) &= \mathcal{T}(\hat{t}) + \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t})\varphi_i(\hat{x})\phi_j(\hat{z}). \end{aligned} \tag{40}$$

As the tip of the beam is assumed to be isolated, the first derivation of  $\theta(\hat{x}, \hat{z}, \hat{t})$ , with respect to  $x$ , becomes zero ( $\frac{\partial \theta}{\partial \hat{x}}|_{(1, \hat{z}, \hat{t})} = 0$ ). Therefore,  $\frac{\partial M_T}{\partial \hat{x}}|_{(1, \hat{t})} = 0$  and, subsequently, it is concluded that:

$$\hat{W}_{0T}(\hat{x}, \hat{t}) = \frac{L^2}{(EI)_{eq}g_0} M_T(1, \hat{t}) \frac{\hat{x}^2}{2}. \tag{41}$$

To determine  $\hat{M}_T$  (dimensionless thermal moment), primarily, the term  $\hat{F}_T$  should be defined. According to Eq. (11),  $\hat{F}_T$  can be expressed as:

$$\hat{F}_T = h \int_{-1/2}^{1/2} \alpha_m \theta(\hat{x}, \hat{z}, \hat{t}) e^{h(\hat{z}+1/2)(\gamma+\beta)} d\hat{z}. \tag{42}$$

By substituting shape functions of temperature changes into Eq. (42),  $\hat{F}_T$  is obtained:

$$\begin{aligned} \hat{F}_T &= \mathcal{B}_1 \mathcal{T}(\hat{t}) + h\alpha_m \sum_i^N \sum_j^N G_{ij}(\hat{t})\varphi_i(\hat{x}) \\ &\int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z}, \end{aligned} \tag{43}$$

in which:

$$\mathcal{B}_1 = h\alpha_m \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z}. \tag{44}$$

Then, by substituting  $\hat{F}_T$  and shape functions of tem-

perature changes into Eq. (14),  $\hat{M}_T$  takes the following form:

$$\begin{aligned} \hat{M}_T(\hat{x}, \hat{z}, \hat{t}) &= \mathcal{K} \left( \hat{M}_T^{(1)} + \hat{M}_T^{(2)} - \hat{M}_T^{(3)} - \hat{M}_T^{(4)} \right), \\ \mathcal{K} &= \frac{L^2}{(EI)_{eq}g_0} \frac{bE_m h^2}{\hat{A}}, \end{aligned} \tag{45}$$

in which:

$$\begin{aligned} \hat{M}_T^{(1)} &= \mathcal{B}_1 \mathcal{T}(\hat{t}) \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} d\hat{z} = \mathcal{B}_1 \tilde{B} \mathcal{T}(\hat{t}), \\ \hat{M}_T^{(2)} &= h\alpha_m \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} \sum_i^N \sum_j^N G_{ij}(\hat{t})\varphi_i(\hat{x}) \\ &\times \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} d\hat{z}, \\ \hat{M}_T^{(3)} &= \tilde{A} \alpha_m \mathcal{T}(\hat{t}) \int_{-1/2}^{1/2} \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \\ &= \tilde{A} \tilde{C} \alpha_m \mathcal{T}(\hat{t}), \tilde{C} = \int_{-1/2}^{1/2} \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z}, \\ \hat{M}_T^{(4)} &= \tilde{A} \alpha_m \sum_i^N \sum_j^N G_{ij}(\hat{t})\varphi_i(\hat{x}) \int_{-1/2}^{1/2} \\ &\times \phi_j(\hat{z}) \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z}. \end{aligned} \tag{46}$$

In conclusion, substituting Eq. (45) into Eq. (41) and, consequently, substituting Eqs. (39), (40) and (42) into Eqs. (26) and (27) leads to the following error functions:

$$\begin{aligned} &\sum_{k=1}^p \ddot{U}_k(\hat{t})\psi_k(\hat{x}) + \hat{c} \sum_{k=1}^p \dot{U}_k(\hat{t})\psi_k(\hat{x}) + \sum_{k=1}^p U_k(\hat{t})\psi_k^{iv}(\hat{x}) \\ &- \frac{\mathcal{A}_1 V^2}{(1 - \hat{W}_s)^3} \sum_{k=1}^p U_k(\hat{t})\psi_k(\hat{x}) + \left( \sum_{i=1}^n \sum_{j=1}^m \ddot{G}_{ij}(\hat{t})\varphi_i(1) \right. \\ &+ \hat{c} \sum_{i=1}^n \sum_{j=1}^m \dot{G}_{ij}(\hat{t})\varphi_i(1) + \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t})\varphi_i''(x) \\ &- \frac{\mathcal{A}_1 V^2}{(1 - \hat{W}_s)^3} \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t})\varphi_i(1) \Big) \\ &\times \left\{ h\alpha_m \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} \right. \\ &\left. \left( \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} \right) d\hat{z} \right\} \end{aligned}$$

$$\begin{aligned}
 & - \tilde{A}\alpha_m \int_{-1/2}^{1/2} \phi_j(\hat{z}) \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \Big\} \mathcal{K} \frac{\hat{x}^2}{2} \\
 & + \left( \ddot{T}(\hat{t}) + \hat{c}\dot{T}(\hat{t}) - \frac{\mathcal{A}_1 V^2}{(1 - \widehat{W}_s)^3} \mathcal{T}(\hat{t}) \right) \\
 & \times \left( \mathcal{B}_1 \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} d\hat{z} - \tilde{A}\alpha_m \int_{-1/2}^{1/2} \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \right) \mathcal{K} \frac{\hat{x}^2}{2} = \epsilon_1, \tag{47}
 \end{aligned}$$

and:

$$\begin{aligned}
 & \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t}) \varphi_i(\hat{x}) \left( k(\hat{z}) \phi_j''(\hat{z}) + \frac{\partial k(\hat{z})}{\partial \hat{z}} \phi_j'(\hat{z}) \right) \\
 & + \mathcal{A}_2 \sum_{i=1}^n \sum_{j=1}^m G_{ij}(\hat{t}) \varphi_i''(\hat{x}) (k(\hat{z}) \phi_j(\hat{z})) \\
 & + \sum_{i=1}^n \sum_{j=1}^m \dot{G}_{ij}(\hat{t}) \varphi_i(\hat{x}) \\
 & \left[ -\mathcal{A}_3 \left( \rho(\hat{z}) C_\nu(\hat{z}) + \frac{2(v(\hat{z})+1)}{1-2v(\hat{z})} \right) \right. \\
 & \left. E(\hat{z}) \alpha(\hat{z})^2 T_\infty \right) \phi_j(\hat{z}) - \mathcal{A}_6 E(\hat{z}) \alpha(\hat{z}) h \alpha_m \\
 & \times \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} \Big] \\
 & + \sum_{i=1}^n \sum_{j=1}^m \dot{G}_{ij}(\hat{t}) \varphi_i(1) (-\mathcal{A}_4 \hat{z} + \mathcal{A}_5) E(\hat{z}) \alpha(\hat{z}) \mathcal{K} \alpha_m \\
 & \times \left[ -h \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} \right. \\
 & \left. \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} d\hat{z} \right. \\
 & \left. + \tilde{A} \int_{-1/2}^{1/2} \phi_j(\hat{z}) \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \right] \\
 & - \sum_{k=1}^p \dot{U}_k(\hat{t}) \psi_k''(\hat{x}) [(-\mathcal{A}_4 \hat{z} + \mathcal{A}_5) E(\hat{z}) \alpha(\hat{z})] \\
 & - \dot{T}(\hat{t}) \mathcal{A}_3 \left( \rho(\hat{z}) C_\nu(\hat{z}) + \frac{2(v(\hat{z})+1)}{1-2v(\hat{z})} E(\hat{z}) \alpha(\hat{z})^2 T_\infty \right) \\
 & + \dot{T}(\hat{t}) E(\hat{z}) \alpha(\hat{z}) \left[ (-\mathcal{A}_4 \hat{z} + \mathcal{A}_5) \mathcal{K} \right.
 \end{aligned}$$

$$\left. \times (-\mathcal{B}_1 \tilde{B} + \tilde{A} \tilde{C} \alpha_m) - \mathcal{A}_6 \mathcal{B}_1 \right] = \epsilon_2. \tag{48}$$

According to the Galerkin method, the following conditions should be satisfied [31]:

$$\begin{aligned}
 & \int_0^1 \psi_r \epsilon_1 d\hat{x} = 0, \quad r = 1, \dots, p \\
 & \int_0^1 \int_{-1/2}^{1/2} \varphi_q \phi_g \epsilon_2 d\hat{z} d\hat{x}, \quad q = 1, \dots, n, \quad g = 1, \dots, m. \tag{49}
 \end{aligned}$$

By employing the Galerkin method, the following reduced order models of coupled thermo-elastic equations can be obtained as:

$$\begin{aligned}
 & \sum_{k=1}^p \mathcal{Q}_{kr}^{(1)} \ddot{U}_k + \hat{c} \sum_{k=1}^p \mathcal{Q}_{kr}^{(1)} \dot{U}_k + \sum_{k=1}^p \mathcal{Q}_{kr}^{(2)} U_k \\
 & - \sum_{k=1}^p \mathcal{Q}_{lr}^{(3)} U_k + \sum_{i=1}^n \sum_{j=1}^m \mathcal{Q}_{ir}^{(4)} \mathcal{Q}_j^{(5)} \ddot{G}_{ij} \\
 & + \hat{c} \sum_{i=1}^n \sum_{j=1}^m \mathcal{Q}_{ir}^{(4)} \mathcal{Q}_j^{(5)} \dot{G}_{ij} + \sum_{i=1}^n \sum_{j=1}^m \mathcal{Q}_{ir}^{(6)} \mathcal{Q}_j^{(5)} G_{ij} \\
 & - \sum_{i=1}^n \sum_{j=1}^m \mathcal{Q}_{ir}^{(7)} \mathcal{Q}_j^{(5)} G_{ij} + \mathcal{Q}_r^{(8)} \mathcal{Q}_j^{(9)} \dot{T}(\hat{t}) \\
 & + \hat{c} \mathcal{Q}_r^{(8)} \mathcal{Q}_j^{(9)} \dot{T}(\hat{t}) - \mathcal{Q}_r^{(10)} \mathcal{Q}_j^{(9)} \mathcal{T}(\hat{t}) = 0, \tag{50} \\
 & \sum_{i=1}^n \sum_{j=1}^m \mathcal{R}_{iq}^{(1)} \mathcal{R}_{jg}^{(2)} G_{ij} + \sum_{i=1}^n \sum_{j=1}^m \mathcal{R}_{iq}^{(3)} \mathcal{R}_{jg}^{(4)} G_{ij} \\
 & + \sum_{i=1}^n \sum_{j=1}^m \mathcal{R}_{iq}^{(1)} \mathcal{R}_{jg}^{(5)} \dot{G}_{ij} + \sum_{i=1}^n \sum_{j=1}^m \mathcal{R}_{iq}^{(6)} \mathcal{R}_{jg}^{(7)} \dot{G}_{ij} \\
 & - \sum_{i=1}^n \sum_{j=1}^m \mathcal{R}_{kq}^{(8)} \mathcal{R}_{jg}^{(9)} \dot{U}_k + \mathcal{R}_q^{(10)} \mathcal{R}_{jg}^{(11)} \dot{T}(\hat{t}) = 0, \tag{51}
 \end{aligned}$$

in which  $\mathcal{Q}$ 's and  $\mathcal{R}$ 's are given in the Appendix.

#### 4. Numerical approach and discussion

In order to apply the Galerkin method to the coupled thermoelastic equations, the following shape functions for a cantilever beam can be considered, which satisfy the boundary conditions (Eq. (37)) [33]:

$$\begin{aligned}
 \psi_k(\hat{x}) & = (\cos h(\Lambda_i \hat{x}) - \cos(\Lambda_i \hat{x})) \\
 & - \alpha_i (\sin h(\Lambda_i \hat{x}) - \sin(\Lambda_i \hat{x})), \tag{52}
 \end{aligned}$$



in which:

$$\alpha_i = \frac{\cos h(\Lambda_i) - \cos(\Lambda_i)}{\sin h(\Lambda_i) - \sin(\Lambda_i)},$$

$$\Lambda_i = 1.8751, 4.6941, 7.8547, 10.958. \tag{53}$$

And the considered shape functions for the heat distribution in  $x$  and  $z$  directions that satisfy the boundary conditions (Eq. (38)) are:

$$\varphi_i(\hat{x}) = \sin\left(\frac{i\pi}{2}\hat{x}\right), \tag{54}$$

$$\phi_j(\hat{z}) = \sin(j\pi\hat{z}). \tag{55}$$

Table 1 shows the material and geometrical properties of the investigated micro-beam. By changing the ceramic constituent percentage of the bottom surface, five different types of FG micro-beam are investigated whose characteristics are shown in Table 2.

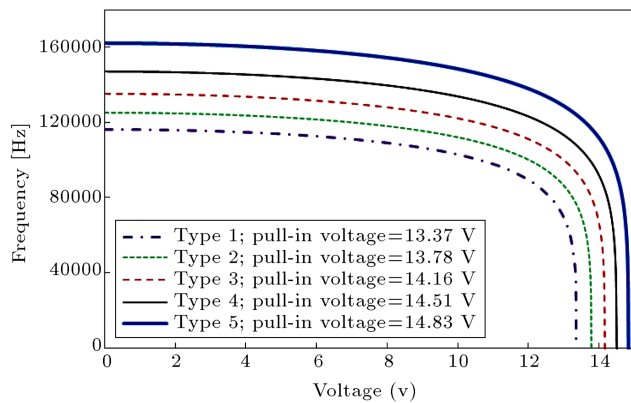
The frequency response of the FG micro-beam depends on the values of the natural frequency of beam and thermal load frequency. In addition, as the applied voltage, as a biasing parameter, changes the beam

**Table 1.** Geometrical, material and air properties of micro-beam.

Symbols	Parameters	Values
$L$	Length	500 $\mu\text{m}$
$B$	Width	90 $\mu\text{m}$
$h$	Thickness	6 $\mu\text{m}$
$g_0$	Initial gap	2 $\mu\text{m}$
$E_m$	Young's modulus	204 Gpa
$\alpha_m$	Thermal expansion	$13.2 \times 10^{-6}\text{K}^{-1}$
$k_m$	Thermal conductivity	Nickel (Ni) $91 \text{ Wm}^{-1}\text{K}^{-1}$
$\rho_m$	Density	(metal) $8908 \text{ kgm}^{-3}$
$C_{\nu m}$	Specific heat	$440 \text{ J}(\text{kg})^{-1}\text{K}^{-1}$
$\nu_m$	Poisson's ratio	0.32
$E_c$	Young's modulus	310 Gpa
$\alpha_c$	Thermal expansion	$3.4 \times 10^{-6}\text{K}^{-1}$
$K_c$	Thermal conductivity	Silicon Nitride ( $\text{Si}_3\text{N}_4$ ) $30 \text{ Wm}^{-1}\text{K}^{-1}$
$\rho_c$	Density	(ceramic) $3300 \text{ kgm}^{-3}$
$C_{\nu c}$	Specific heat	$711 \text{ J}(\text{kg})^{-1}\text{K}^{-1}$
$\nu_c$	Poisson's ratio	0.24
$\varepsilon_0$	Permittivity of air	8.854 pF/m
$T_\infty$	Temperature of air	298 K

**Table 2.** Characteristics of five several types of FG micro-beams.

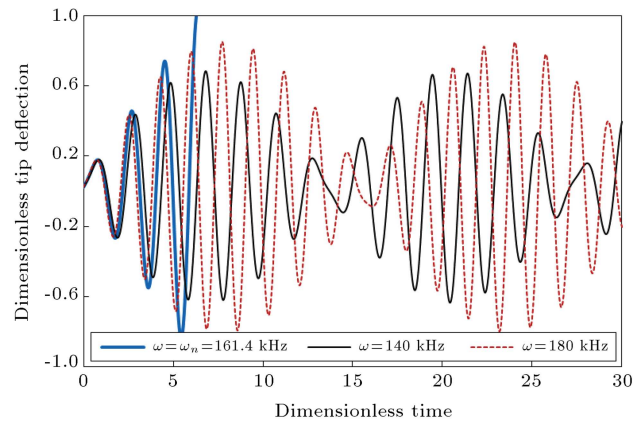
Type	1	2	3	4	5
Ceramic percent of bottom surface	0% (metal-rich)	25%	50%	75%	100%
$E_b$ (Gpa)	204	230.5	257	283.5	310
$\alpha_b \times 10^{-6}\text{K}^{-1}$	13.2	10.75	8.3	5.85	3.4
$k_b$ ( $\text{Wm}^{-1}\text{K}^{-1}$ )	90.9	75.65	60.5	45.2	30
$\rho_b$ ( $\text{kgm}^{-3}$ )	8908	7506	6104	4702	3300
$C_{\nu b}$ ( $\text{J}(\text{kg})^{-1}\text{K}^{-1}$ )	4400	3477.8	2555.5	1633.3	711.8
$\nu_b$	0.32	0.3	0.28	0.26	0.24
$\gamma$	0	20355	48493	54849	69742
$\beta$	0	-34218.5	-77327	-135629	-226073
$\lambda$	0	-30552	-67991	-116352	-187460
$\mu$	0	-28541	-63001	-106490	-165500
$\eta$	0	-39191	-90559	-165170	-303590
$S$	0	-10756	-22255	-34607	-47947



**Figure 2.** Variation of the first natural frequency for five types of FG micro-beam versus DC voltage.

natural frequency, the natural frequency of the FG micro-beam versus the bias DC voltage is determined and illustrated in Figure 2. The figure shows the non-dimensional first natural frequency for the five types of FG micro-beam versus rise in primal voltage. Due to an increase in the equivalent stiffness of the beam, as shown in Figure 2, enhancing the ceramic constituent percentage of the FG beam increases the value of the first natural frequency. In addition, it reveals that the value of the natural frequency is decreased by increasing the applied DC voltage until static pull-in phenomenon occurs and the natural frequency of the micro-beam becomes zero. Therefore, the static pull-in instability is a kind of stationary instability [23]. When the applied voltage exceeds the pull-in value, the stable equilibrium positions of the micro beam cease to exist. This is unwanted because it leads the structure to collapse and, consequently, failure in the devices. Pull-in instability significantly limits the stable range of the operation of the capacitive micro beams.

To compare the obtained pull-in voltages with the results of references, a simple and homogeneous micro-beam, similar to the first type, is used (see Table 3). The parameters used in this simple homogenous micro-



**Figure 3.** Time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip for different thermal excitation frequencies (primal voltage = 5 V).

beam are  $E = 169$  (GPa),  $75 \leq L \leq 250$  ( $\mu\text{m}$ ),  $h = 2.94$  ( $\mu\text{m}$ ),  $b = 50$  ( $\mu\text{m}$ ) and  $g_o = 1.05$  ( $\mu\text{m}$ ).

Figure 3 declares the dimensionless end deflection of the 5th type of FG micro-beam versus time for different thermal excitation frequencies, when it is deflected owing to the application of 5 [V] primal bias voltage. Hereafter, the 5th type is the reference type whenever no type is specified. As shown in Figure 3, the response amplitude of the beam, as expected, grows unboundedly when the thermal excitation frequency equals the natural frequency of the beam. Resonance phenomenon in capacitive micro beams causes an increase in the amplitude of the response. Of course, it must be noted that the electrostatic force is a displacement dependent force. Therefore, increasing the amplitude of the response can lead the structure to an unstable condition through a homoclinic bifurcation [23]. However, because, in this work, results are obtained through a linearized equation, this phenomenon cannot be seen. Another point that deserves attention is that, as anticipated, when the excitation frequency is close, but not exactly equal,

**Table 3.** Pull-in voltages of homogenous cantilever micro-beams: comparisons with experimentally measured data.

Cantilever length ( $\mu\text{m}$ )	Pull-in voltages		Error, $\Delta$ (%)
	in proposed model (V)	Pull-in voltage based on the experimental observations [34], (V)	
75	77.43	76.2	1.6
100	43.56	43.5	0.1
125	27.88	28.1	0.7
150	19.36	19.6	1.2
175	14.23	14.5	1.9
200	10.9	10.9	0
225	8.61	9.3	7.4
250	6.97	7.3	4.5

to the natural frequency, the “*beating*” phenomenon occurs.

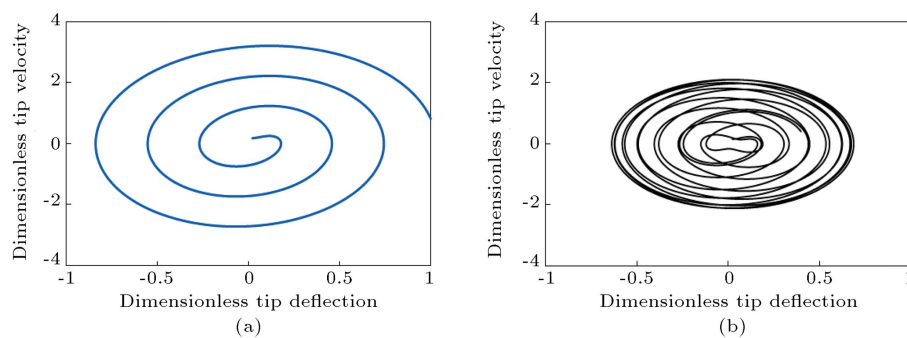
Phase portraits reveal many important characteristics about the dynamic behavior of the system, i.e. specify that the solution of the system is stable or unstable. Figure 4(a) and (b) demonstrate the phase portrait of the 5th type of the FG micro-beam. When the thermal excitation frequency is equal to the first natural frequency of the beam, the amplitude increases without bound as long as the tip of the beam contacts the substrate (Figure 4(a)). As mentioned earlier, in a non-linearized system, contact with the substrate will be occurred through a homoclinic bifurcation. Figure 4(b) shows the dimensionless phase portrait of a bounded response for the 5th type of FG micro-beam when the thermal excitation frequency is 1404 kHz.

Figure 5(a)-(d) illustrate the dimensionless end deflection of the 5th type of FG micro-beam versus

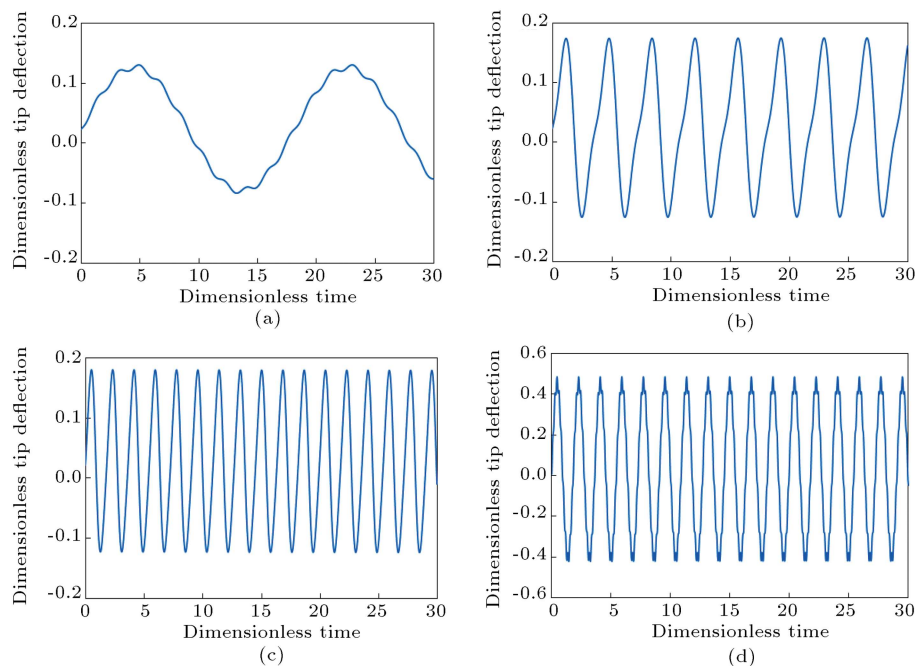
time when various thermal excitation frequencies are applied.

In Figure 6, the dimensionless end deflection of the different types of FG micro-beam versus time is shown. From this data, it can be resulted that FG beams with lower ceramic constituent percentage, contact later with the stationary electrode. The reason is, the lower the ceramic constituent percentage of the FG micro-beam, the lower the amount of thermal moment.

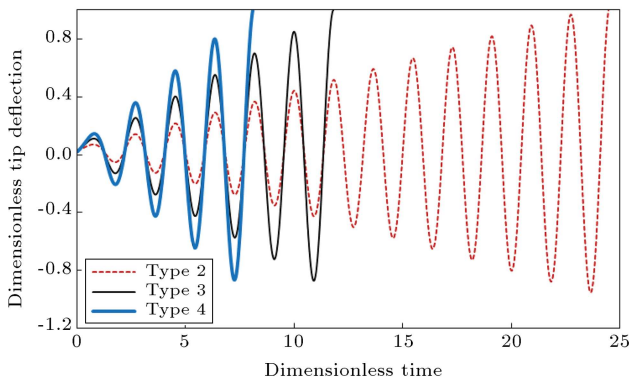
As the value of the internal thermo-elastic damping is small and the free vibration response can be seen in the responses for long periods of time, Figures 7 and 8 present oscillatory motion, considering an external equivalent coefficient of damping, such as squeeze film damping. Figure 7 shows the time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip when the applied thermal excitation frequency is considered equal with



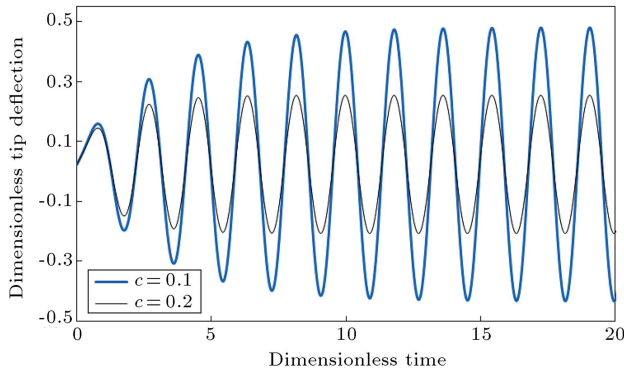
**Figure 4.** Dimensionless phase portraits: a)  $\omega = \omega_n$ ; and b)  $\omega = 140$  kHz.



**Figure 5.** Time-history of the dimensionless transverse deflection of the 5th of FG micro-beam tip: a)  $\omega = 0.1 \omega_n$ ; b)  $\omega = 0.5 \omega_n$ ; c)  $\omega = 2 \omega_n$ ; and d)  $\omega = 10 \omega_n$  (primal voltage = 5 V).



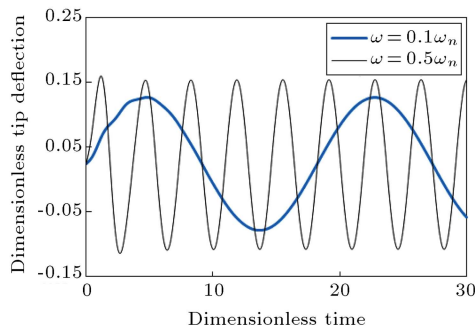
**Figure 6.** Time-history of the dimensionless transverse deflection of different types of FG micro-beam tip in the resonance case (primal voltage = 5 (V)).



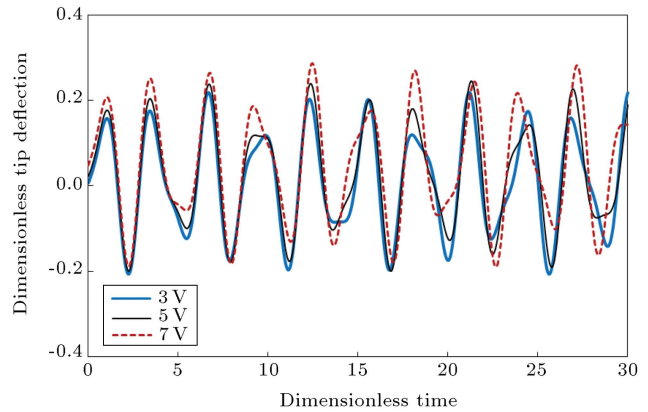
**Figure 7.** Time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip when  $\omega = \omega_n$  and various external damping coefficients are considered (primal voltage = 5 V).

the natural frequency. It is apparent from this figure that when the damping coefficient is increased, the amplitude of vibration decreased.

Figure 8(a) and (b) show time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip for different thermal excitation frequencies, considering an equivalent external damping coefficient ( $c = 0.1$ ) with a primal bias DC



**Figure 8.** Time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip when different thermal excitation frequencies are applied considering arbitrary external damping coefficient ( $c = 0.1$ , primal voltage = 5 V).



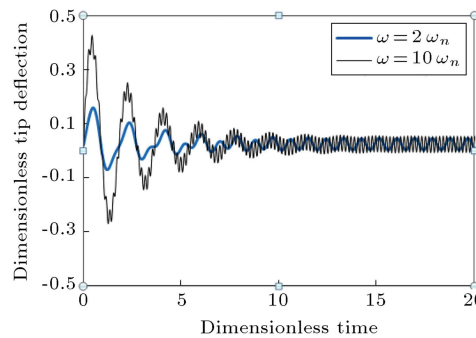
**Figure 9.** Time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip when different primal voltages are applied ( $\omega = 100$  kHz).

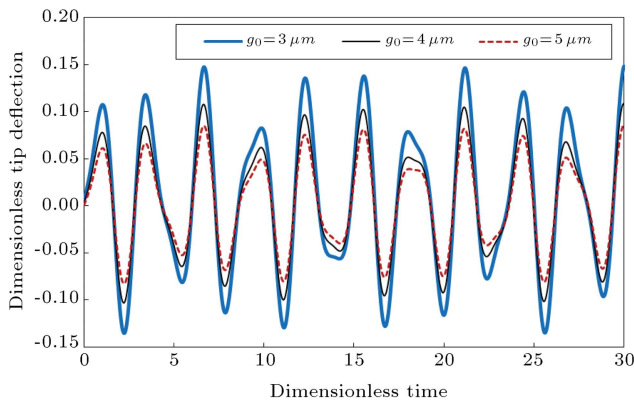
voltage equal to 5 (V). According to Figure 8, as time passes, the free vibration response is dissipated and the FG beam is oscillated with the thermal excitation frequency and constant amplitude.

Figure 9 depicts the effect of applied bias DC voltage on the time-history response of the 5th type of FG micro-beam tip. From the figure, it can be concluded that by increasing the applied bias DC voltage, not only is the static equilibrium position of the beam raised, but also the amplitude of the response (100 kHz arbitrary thermal excitation frequency is applied.)

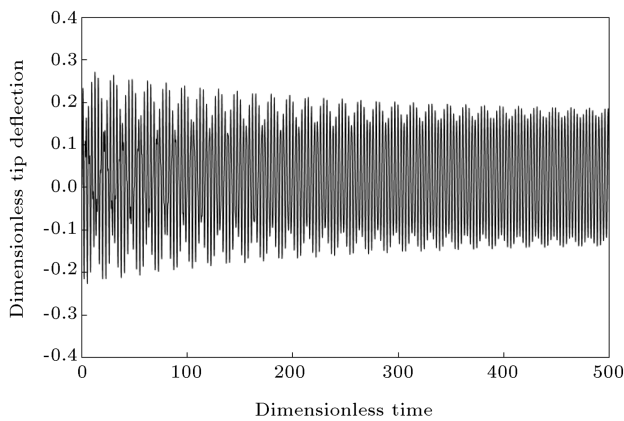
The effect of changing initial gap ( $g_0$ ) on the time history response of the 5th type of FG micro-beam tip is presented in Figure 10. The data would seem to suggest that increasing the amount of initial gap causes a decrease in both static equilibrium position and amplitude of response. It is clear from Eq. (16) that electrostatic force will be decreased when the initial gap is increased.

Because the thermo-elastic coupling term is not usually distinguishable with the uncoupled solution for different materials [15], Figure 11 is given to show the time history of the dimensionless transverse deflection





**Figure 10.** Time-history of the dimensionless transverse deflection of the 5th type of FG micro-beam tip when various initial gaps are considered ( $\omega = 100$  kHz).

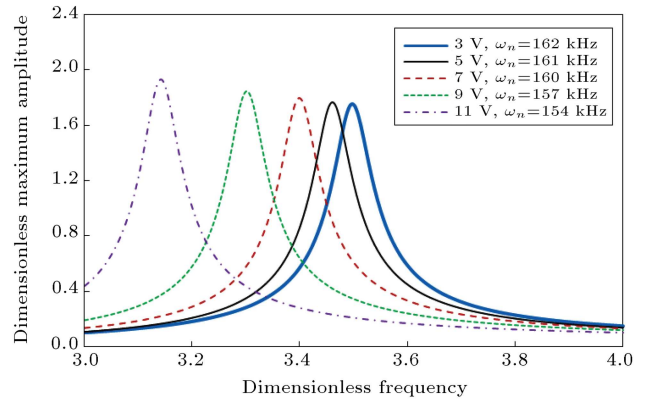


**Figure 11.** Time-history of the dimensionless transverse deflection of the 5th of FG micro-beam tip, the magnified coupling effect (primal voltage = 5 V and  $\omega = 100$  kHz).

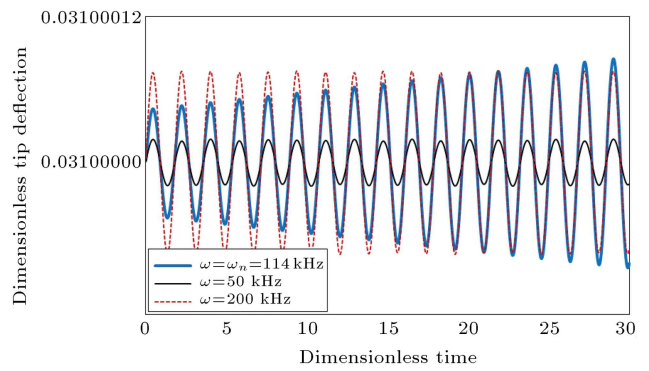
of the 5th type of FG micro-beam tip when the coupling term between strain and temperature is magnified 1000 times. It is observed that the amplitude of vibration decreases with time.

Figure 12 shows the dimensionless maximum amplitude of the 5th type of FG micro-beam, with respect to dimensionless frequency, when various voltages are applied. As mentioned previously, Figure 12 indicates that increasing the applied bias voltage causes a decrease in the natural frequency and, on the contrary, an increase in the amplitude of vibration. Note that dimensionless frequency is introduced as  $\hat{\omega}$ , and in this study, it equals  $\hat{\omega} = \omega t^*$ . The magnitude of  $t^*$ , considering Eq. (25), is obtained as  $2.14 \times 10^{-5}$ .

Figure 13 illustrates the dimensionless time-history response for the homogenous micro-beam (metal rich) when it is deflected electrostatically by 5 V bias DC voltage. For the homogenous beam, the thermal moment equation ( $M_T$ ) is not included,  $\mathcal{T}(\hat{t})$ . This difference between the thermal moments of the FG and homogenous beams leads to different responses to the thermal actuation. By comparing the responses of



**Figure 12.** Frequency-response curves of 5th type of a FG micro-beam when various primal voltages are applied.



**Figure 13.** Time-history of the dimensionless transverse deflection of the homogenous micro-beam tip for different thermal excitation frequencies (primal voltage = 5 V).

the FG and homogenous beams, it can be concluded that the vibration amplitudes of the FG beam are significantly greater than those of the homogenous beam. This is because, in the latter, thermal moment is created only due to a small temperature gradient related to the thermo-elastic coupling effect.

$$\widehat{M}_T = \mathcal{K} \left( -\widehat{M}_T^{(4)} \right) = -\mathcal{K} \tilde{A} \alpha_m$$

$$\sum_i^N \sum_j^N G_{ij}(\hat{t}) \varphi_i(\hat{x}) \int_{-\frac{1}{2}}^{\frac{1}{2}} \phi_j(\hat{z}) \hat{z} d\hat{z}. \quad (56)$$

### 5. Conclusions

The objective of this paper is to present the dynamic response of an electrostatically deflected capacitive FG cantilever micro-beam to harmonically varying thermal loads. Vibration is incited due to the different material properties of the beam constituents and the thermo-elastic coupling effect. Boundary conditions of the FG micro-beam are taken to be cantilever, with the tip to be isolated. Results describe the dynamic transverse deflection and frequency response curves of the FG micro-beam. In this study, the FG micro-beam is

made of a mixture of metal and ceramic, where the thermo-mechanical properties of the FG micro-beam are all assumed to vary exponentially through the thickness. The theoretical governing coupled thermo-elastic formulations are based on the Euler-Bernoulli beam theory and infinite speed of heat propagation. To solve the problem, the Galerkin formulation method is used. Results illustrate the effect of variation of bias DC voltage and changing ceramic constituent percentage on the FG micro-beam time-history and frequency responses. The numerical calculation demonstrates that increasing the ceramic constituent percent of the beam material causes the beam tip to contact with stationary electrodes in less periods of time. This is due to increasing the difference of the thermal expansion coefficient between the FG micro-beam surfaces, which results in an increase in thermal moment. It is also shown that the vibration amplitudes of the FG micro-beam are significantly greater than those of the homogenous beam, since, in the latter, the thermal moment is created only because of a small temperature gradient owing to the thermo-elastic coupling effect. Moreover, the effect of thermo-elastic damping on the dynamic responses of the beam is negligible.

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**Appendix**

The parametrs  $Q$  and  $R$  appeared in the text are listed below:

$$Q_{kr}^{(1)} = \int_0^1 \psi_k \psi_r d\hat{x}, \quad Q_{kr}^{(2)} = \int_0^1 \psi_k^{iv} \psi_r d\hat{x},$$

$$Q_{kr}^{(3)} = \int_0^1 \frac{A_1 V^2 \psi_k}{(1 - \hat{W}_s)^3} \psi_r d\hat{x},$$

$$Q_{ir}^{(4)} = \int_0^1 \varphi_i(1) \frac{\hat{x}^2}{2} \psi_r d\hat{x},$$

$$Q_j^{(5)} = \mathcal{K} \alpha_m \left( h \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} d\hat{z} - \tilde{A} \int_{-1/2}^{1/2} \phi_j(\hat{z}) \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \right),$$

$$Q_{ir}^{(6)} = \int_0^1 \varphi_i'' \frac{\hat{x}^2}{2} \psi_r d\hat{x},$$

$$Q_{ir}^{(7)} = \int_0^1 \frac{A_1 V^2}{(1 - \hat{W}_s)^3} \varphi_i(1) \frac{\hat{x}^2}{2} \psi_r d\hat{x},$$

$$Q_r^{(8)} = \int_0^1 \frac{\hat{x}^2}{2} \psi_r d\hat{x},$$

$$Q_j^{(9)} = \mathcal{K} \left( \mathcal{B}_1 \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} d\hat{z} - \tilde{A} \alpha_m \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \right),$$

$$Q_r^{(10)} = \int_0^1 \frac{A_1 V^2}{(1 - \hat{W}_s)^3} \hat{x}^2 \psi_r d\hat{x},$$

$$\mathcal{R}_{iq}^{(1)} = \int_0^1 \varphi_i \varphi_q d\hat{x},$$

$$\mathcal{R}_{jg}^{(2)} = \int_{-1/2}^{1/2} \left( k(\hat{z}) \phi_j'' + \frac{\partial k(\hat{z})}{\partial \hat{z}} \phi_j' \right) \phi_g d\hat{z},$$

$$\mathcal{R}_{iq}^{(3)} = \int_0^1 \varphi_i'' \varphi_q d\hat{x},$$

$$\mathcal{R}_{jg}^{(4)} = A_2 \int_{-1/2}^{1/2} (k(\hat{z}) \phi_j) \phi_g d\hat{z},$$

$$\mathcal{R}_{iq}^{(6)} = \int_0^1 \varphi_i(1) \varphi_q d\hat{x},$$

$$\mathcal{R}_{kq}^{(8)} = \int_0^1 \psi_k'' \varphi_q d\hat{x},$$

$$\mathcal{R}_{jg}^{(9)} = \int_{-1/2}^{1/2} (-A_4 \hat{z} + A_5) E(\hat{z}) \alpha(\hat{z}) \phi_g d\hat{z},$$

$$\mathcal{R}_q^{(10)} = \int_0^1 \varphi_q d\hat{x},$$

$$\mathcal{R}_{jg}^{(5)} = \int_{-1/2}^{1/2} \left[ -\mathcal{A}_3 \left( \rho(\hat{z}) C_\nu(\hat{z}) + \frac{2(v(\hat{z}) + 1)}{1 - 2v(\hat{z})} E(\hat{z}) \alpha(\hat{z})^2 T_\infty \right) \phi_j(\hat{z}) - \mathcal{A}_6 E(\hat{z}) \alpha(\hat{z}) h \alpha_m \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} \right] \phi_g d\hat{z},$$

$$\mathcal{R}_{jg}^{(7)} = \int_{-1/2}^{1/2} (-\mathcal{A}_4 \hat{z} + \mathcal{A}_5) E(\hat{z}) \alpha(\hat{z}) \mathcal{K} \alpha_m \left( -h \int_{-1/2}^{1/2} \hat{z} e^{\gamma h(\hat{z}+1/2)} \int_{-1/2}^{1/2} e^{(\gamma+\beta)h(\hat{z}+1/2)} \phi_j(\hat{z}) d\hat{z} d\hat{z} + \tilde{A} \int_{-1/2}^{1/2} \phi_j(\hat{z}) \hat{z} e^{(\gamma+\beta)h(\hat{z}+1/2)} d\hat{z} \right) \phi_g d\hat{z},$$

$$\mathcal{R}_{jg}^{(11)} = \int_{-1/2}^{1/2} \left[ -\mathcal{A}_3 \left( \rho(\hat{z}) C_\nu(\hat{z}) + \frac{2(v(\hat{z}) + 1)}{1 - 2v(\hat{z})} E(\hat{z}) \alpha(\hat{z})^2 T_\infty \right) + E(\hat{z}) \alpha(\hat{z}) [(-\mathcal{A}_4 \hat{z} + \mathcal{A}_5) \mathcal{K}(-\mathcal{B}_1 \tilde{B} + \tilde{A} \tilde{C} \alpha_m) - \mathcal{A}_6 \mathcal{B}_1] \right] \phi_g d\hat{z}.$$

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